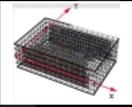
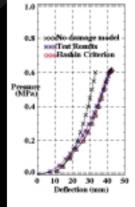
## Rotoral Engineering & Design Centre **Progressive Failure Analysis of** Laminated Composite **Structures**

C E D





Load Central Deflection Graph. Aspect Ratio 1

However, about 40000 load steps are needed to satisfy equilibrium at each damaged material point, which requires CPU time of the order of days. Therefore the use of sequential programs becomes inadequate for such an analysis. The use of a parallel FE program ABAQUS on an IBM SP2

is currently being investigated

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Matrix Crack at 0.10 MPa and fibre Matrix Shear at Final Failure. Aspect Ratio 2

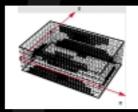
The progressive damage model includes all the well established strength theories for failure detection.It predicts the load deformation behaviour better than that of the no-damage model.

extensively used in maritime

Composite materials are

structures.

But the concept of progressive damage of laminated composites requires thousands of load steps for a safe and reliable assessment of their strength at collapse.



Matrix Cracks at 0.10 MPa

and fibre Breakage at Final

Failure. Aspect Ratio 1

The Laminate Boundary Value Problem